

Failure Assessment of a Thin-Walled Pipe Subjected to an Overload During Manufacturing

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The presentation is based on a study performed by



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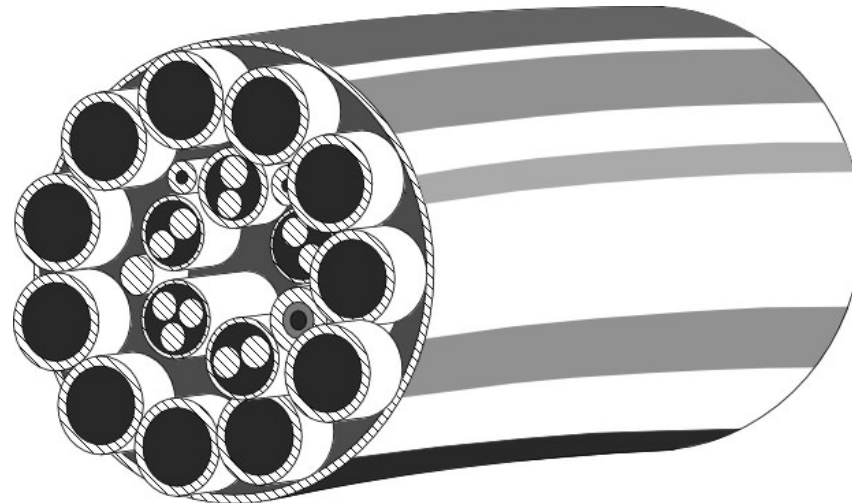


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Subsea umbilicals: composite cable and small diameter tubular bundles deployed on the seabed in conjunction with offshore installations for oil or gas exploitation.

Their purpose: to supply necessary power and control to remote production valves, chokes and control systems.

Loading: alternating internal pressure

Once an umbilical is installed on the seabed failure-free operation is intended over its whole lifetime (safe-life design philosophy).

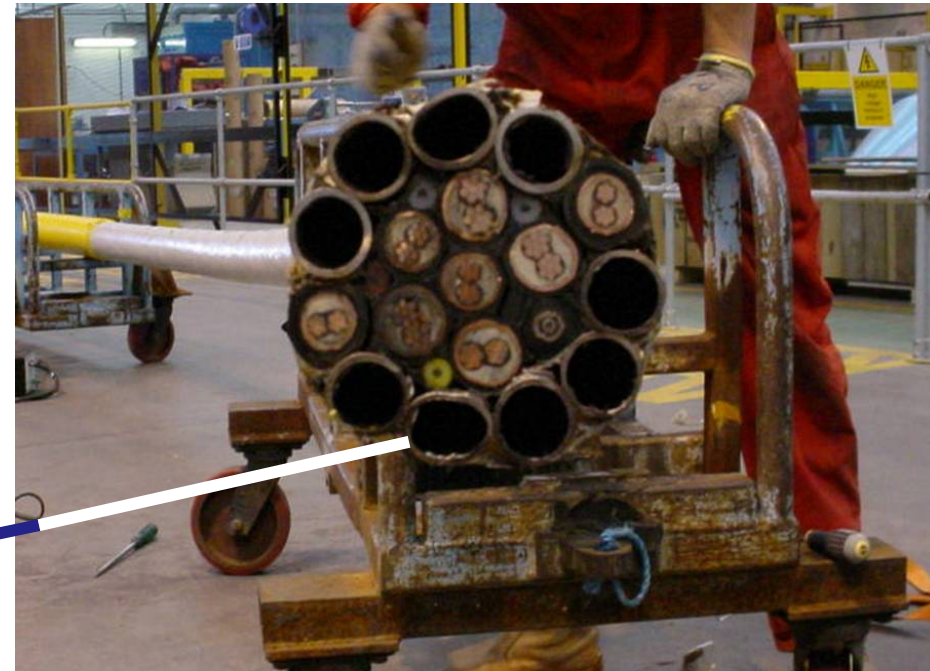
Present case: The performance check revealed that one tube of an umbilical was not able to carry load, i.e., it had a leak.

To find the reason a spare line section was investigated which had experienced the same treatment during manufacture.



Material: SAF 2507

- Yield strength: 756 MPa
- Tensile strength: 947 MPa
- Elongation: 31 %,



It was found that some of the tubes were largely ovalised (to an extent of 55%).

Subsequently the tubes were subjected to internal pressure in order to re-round them.

Problem: It could not be excluded that potential cracks just small enough not to snap through the wall existed in the re-rounded umbilical.

Question: Could such a crack cause the failure of one or more of the tubes within the projected lifetime?



Two-Steps Fracture Mechanics Analysis

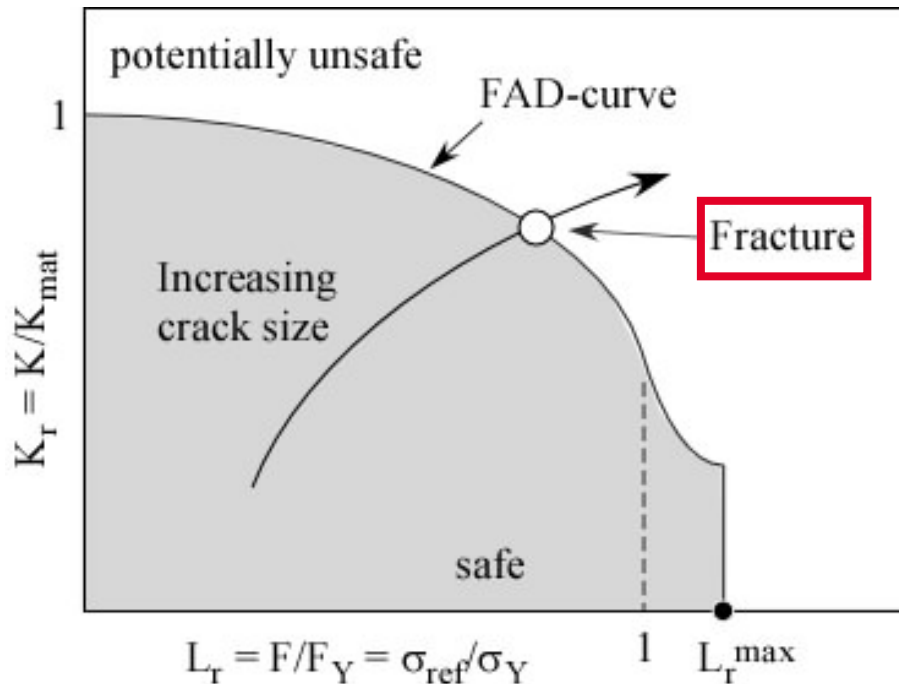
Step 1: Determine the potential maximum crack size that could have survived the re-rounding process.

Step 2: Determine the minimum residual lifetime based on this crack size and compare it with the projected lifetime.

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FAD curve: $K_r = f(L_r)$

$$K_r = K/K_{mat}$$

K – Stress intensity factor (K factor)

K_{mat} – Fracture resistance of the material

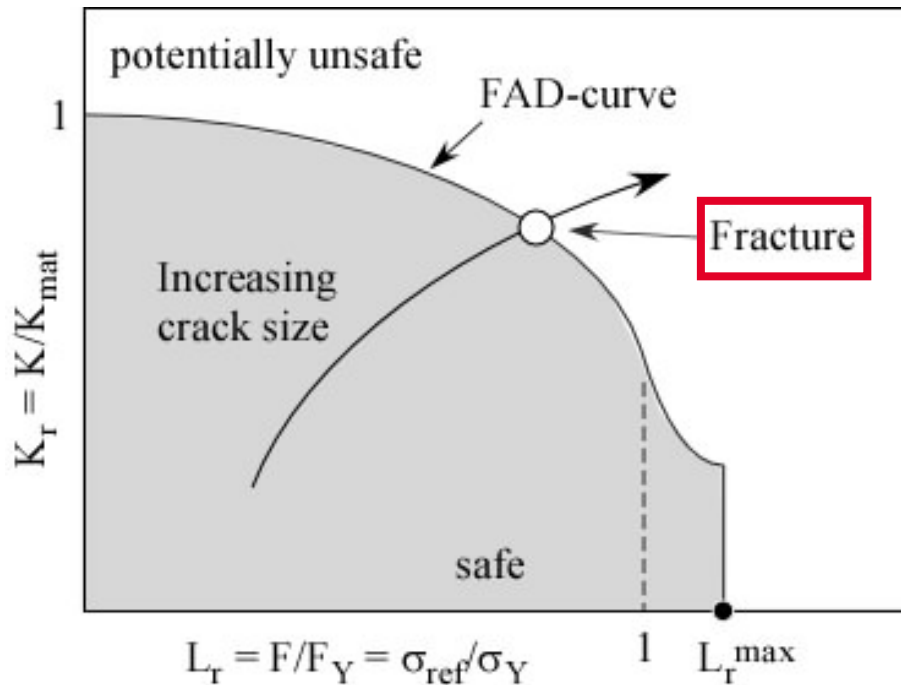
$$L_r = F/F_Y = \sigma_{ref} / \sigma_Y$$

F – Load; F_Y – Yield load

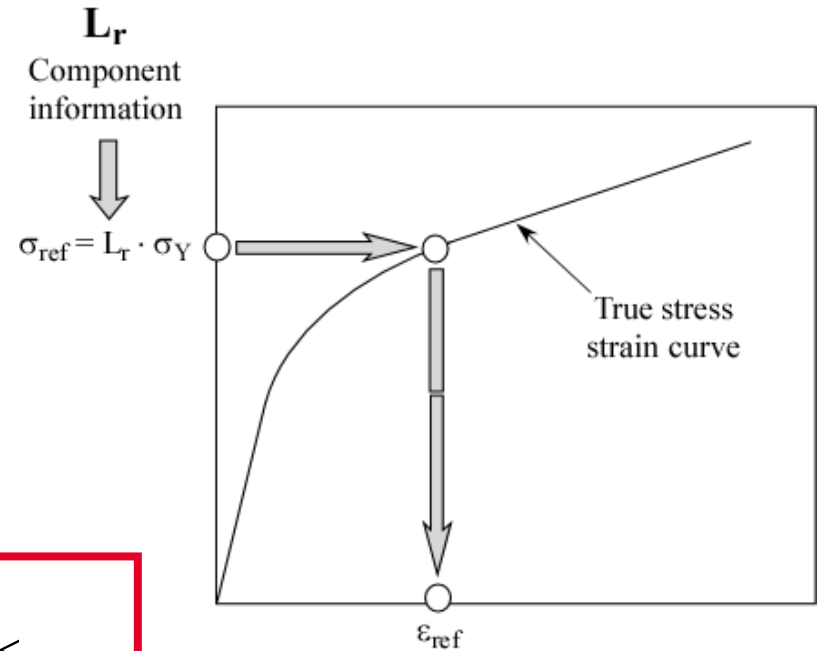
σ_{ref} – (Net section) reference stress

σ_Y – Yield strength of the material

Step 1: Determine the potential maximum crack size that could have survived the re-rounding process.



FAD curve: $K_r = f(L_r)$



$$f(L_r) = \begin{cases} \left[\frac{E \cdot \epsilon_{ref}}{\sigma_{ref}} + \frac{1}{2} \frac{L_r^2}{E \cdot \epsilon_{ref} / \sigma_{ref}} \right]^{-1/2} & \text{for } L_r < L_r^{\max} \\ 0 & \text{for } L_r = L_r^{\max} \end{cases}$$

Usually, in the context of a structural integrity analysis, conservatism means that the the critical crack size is **underestimated**.

However, in the context of overload based analyses, things are different:



In this case conservatism means that the critical crack size has to be **overestimated!**

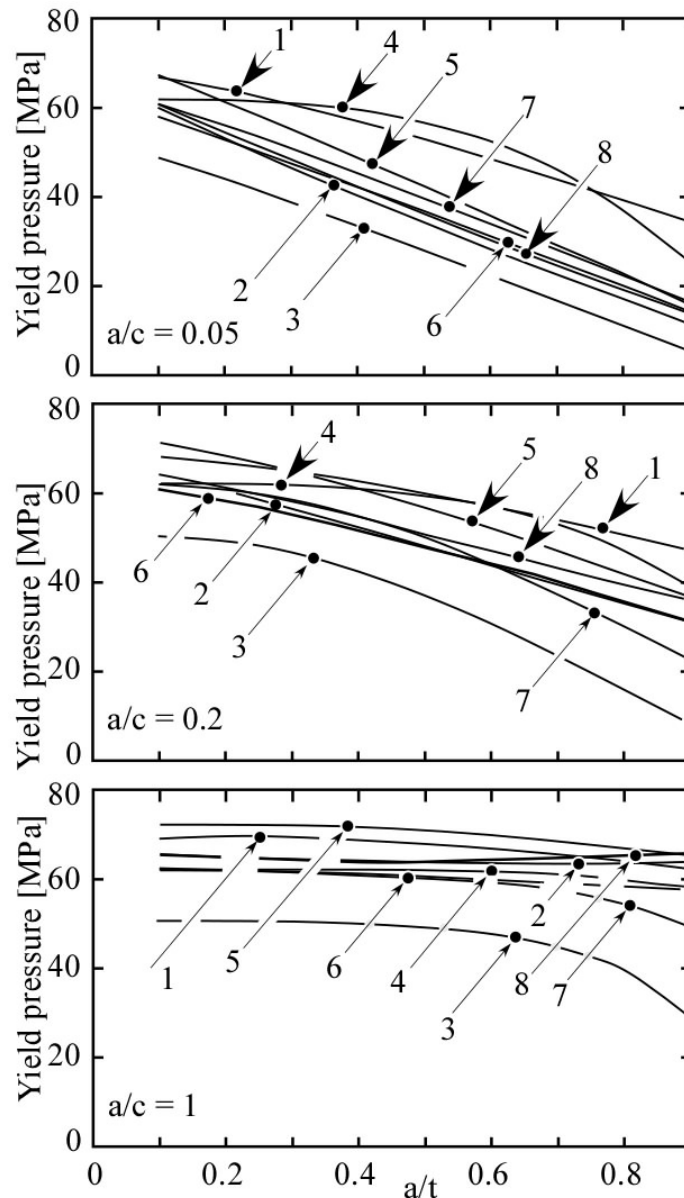
Caution: Whilst, in principle, a fracture mechanics analysis of a tube subjected to internal pressure is state-of-the-art the problem of reversed conservatism is usually not covered.

The relevant parameters are:

the fracture resistance of the material (K_{mat}) and

the net section yield load of the structure with crack (F_Y or σ_{ref})

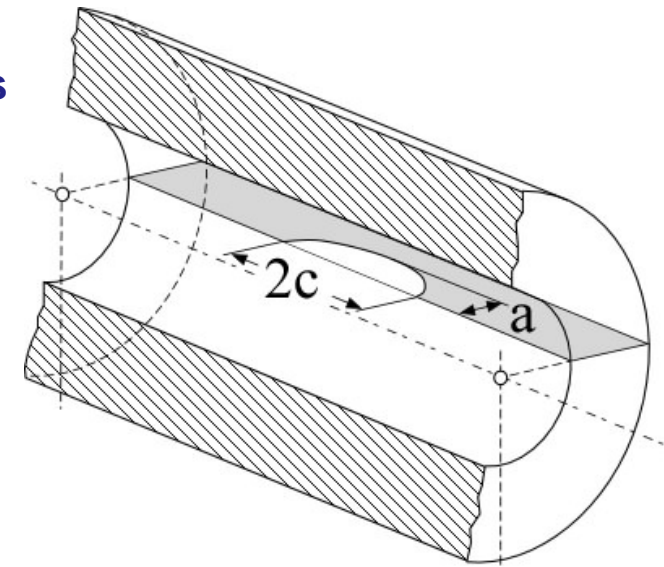
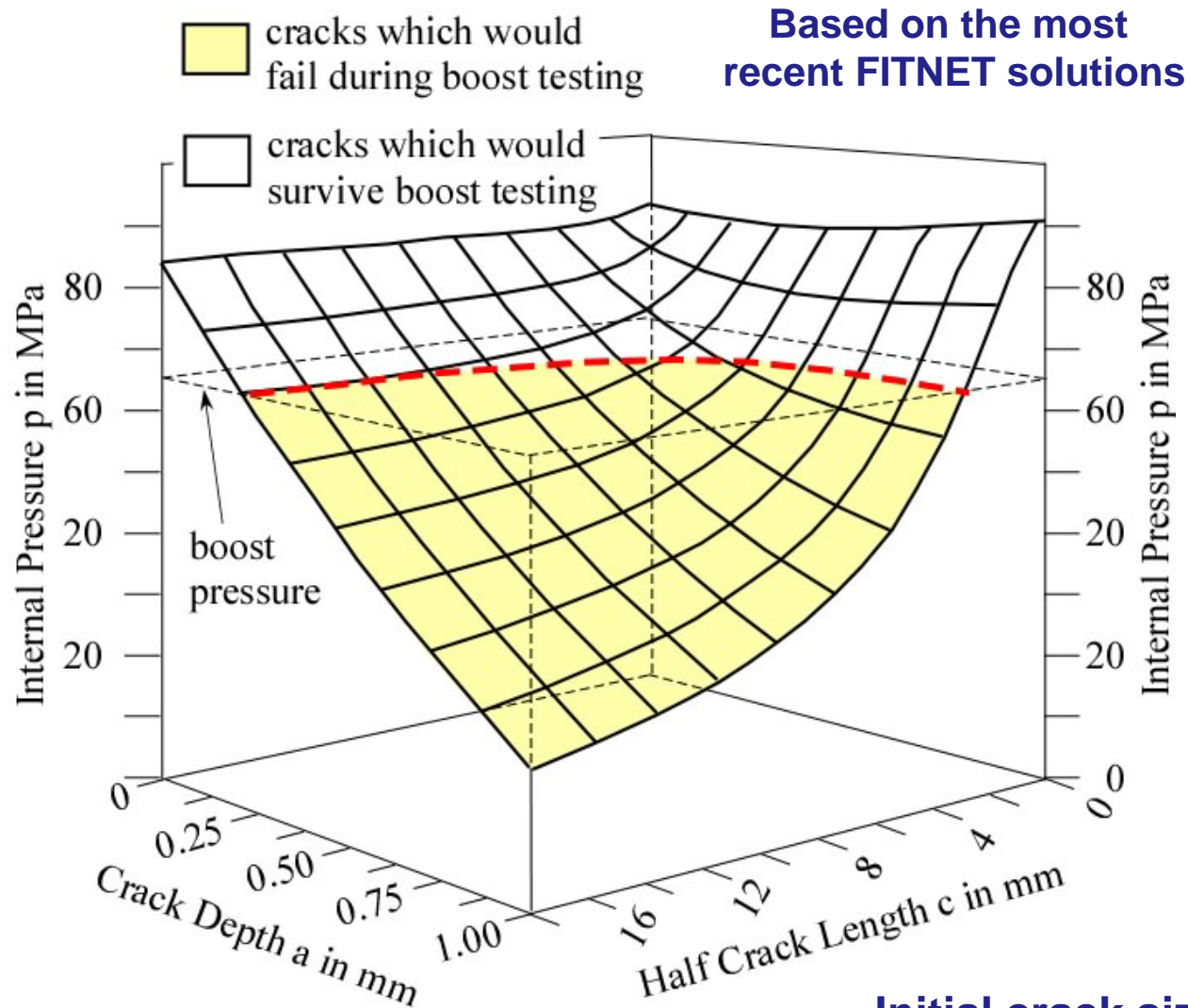
For both parameters upper bound values or solutions are needed.



- (1) Kim & Shim (2005)
- (2) R6, Rev. 4 (2001)
- (3) BS 7910 (2005) (**standard solution**)
- (4) Sattari-Far & Dillström (2004)
- (5) Staat & Khoi Vu (2007)
- (6) API 579-N (2000)
- (7) API 579-L (2000)
- (8) FITNET (2006)

Aspects of different solutions:

- Definition and determination
- Local and global solutions
- Crack face pressure



Dependent on its length at surface ($2c$) the maximum depth (a) of the crack according to the internal pressure at re-rounding is identified.

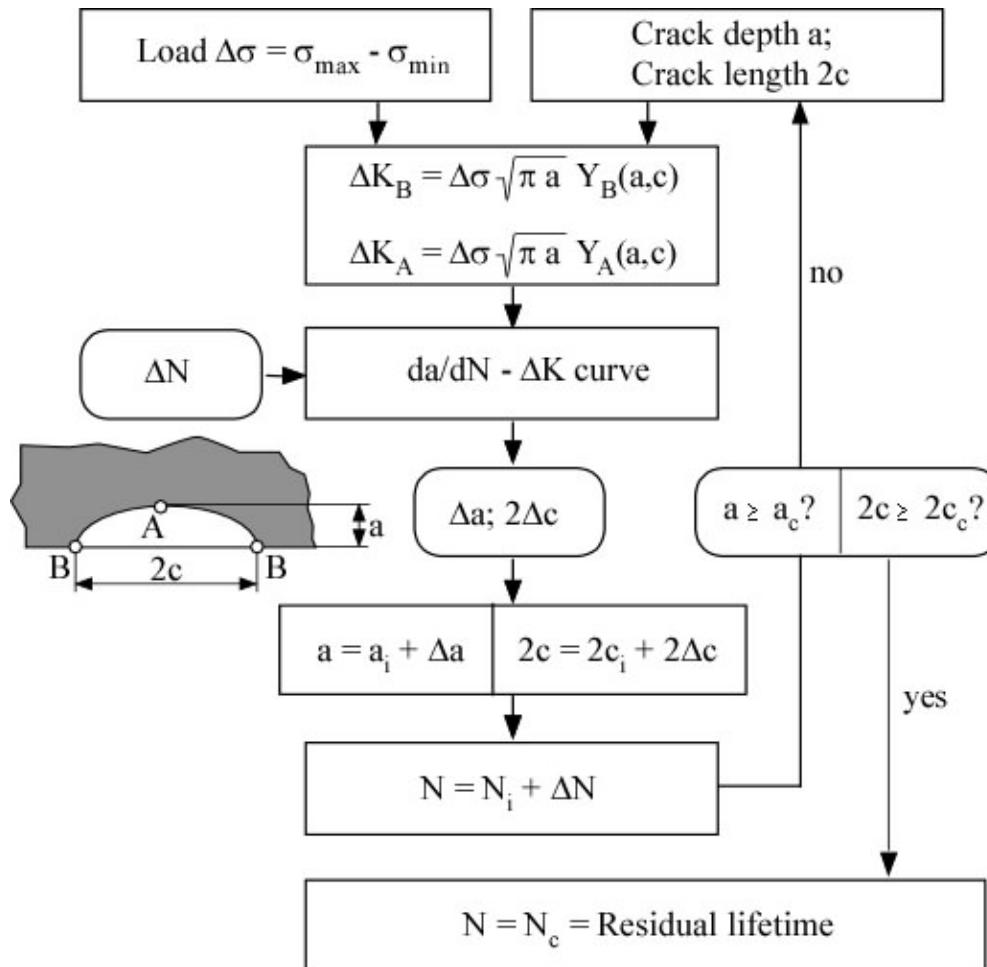


Initial crack size for Step 2 of the analysis

Step 1: Determine the potential maximum crack size that could have survived the re-rounding process.

Step 2: Determine the minimum residual lifetime based on this crack size and compare it with the projected lifetime.

Step 2: Determine the residual lifetime of the maximum crack of Step 1 under in-service conditions.



Loading:

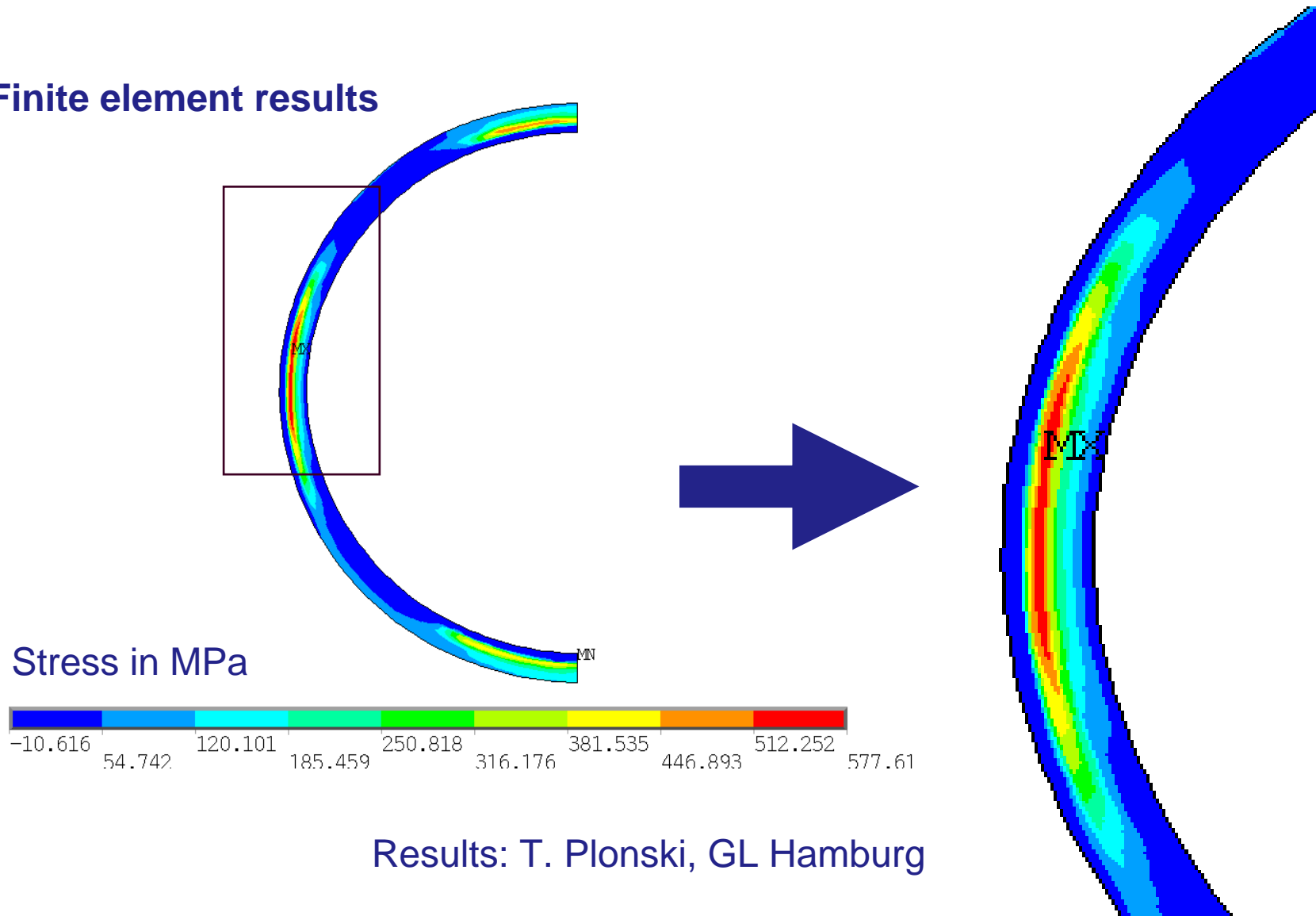
Alternating internal pressure in-service

Caution:

1) Conservatism turns back to normal!

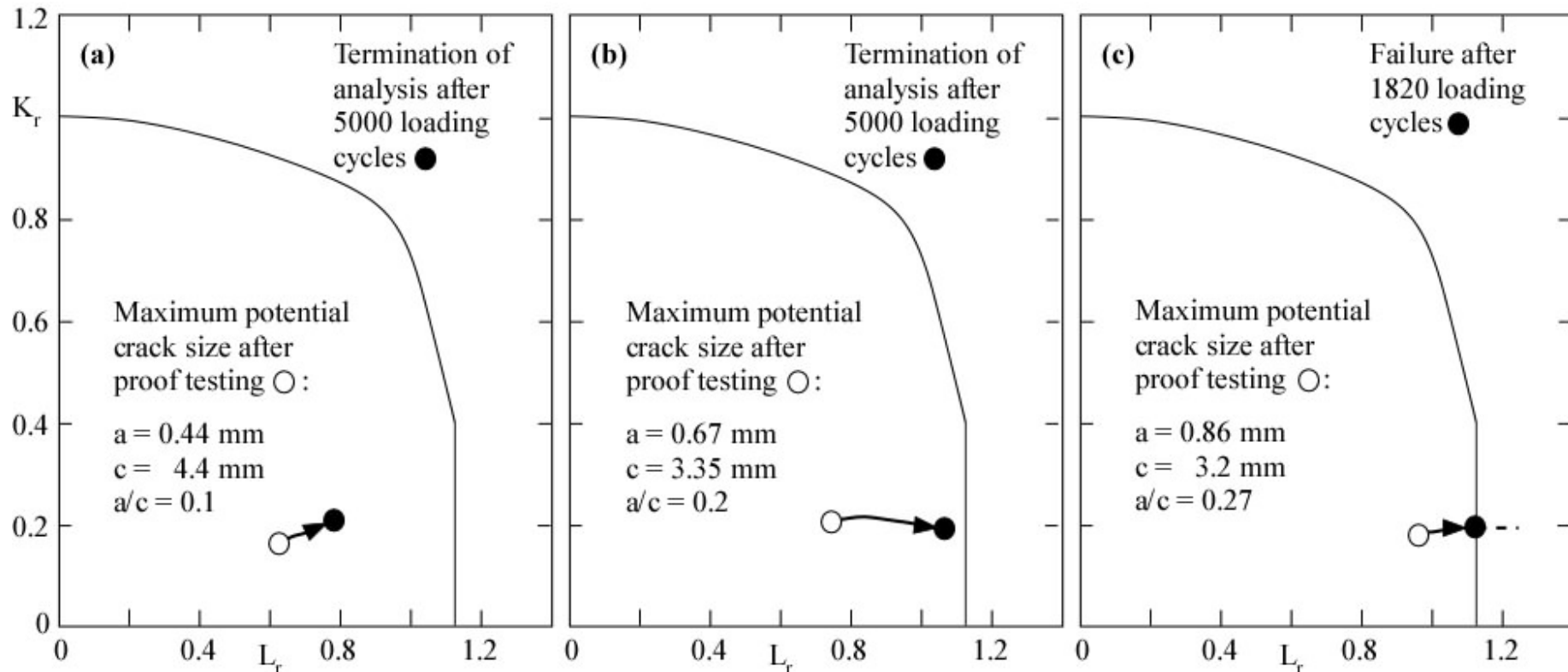
2) The residual stresses due to the re-rounding process have to be taken into account as an additional loading component!

Finite element results



Results: T. Plonski, GL Hamburg

Three possible maximum crack dimensions after re-rounding:

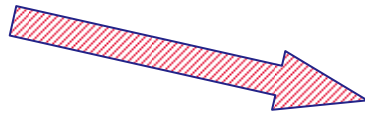


**Termination after 5000 loading cycles
(no failure)**

**Most critical case
(failure after 1820 cycles)**

- **The critical crack dimensions $a = 0.86$ mm and $c = 3.2$ mm after re-rounding yielded final failure after only 1820 loading cycles which is less than the projected lifetime.**
- **Therefore, additional measures were necessary (e.g., refined non-destructive inspection or additional tests on the spare line sections)**
- **Note that the analysis followed a worst case scenario, i.e., it was based on a number of pessimistic assumptions:**
 - (a) **Even little increase in pressure would have yielded failure already during re-rounding.**
 - (b) **The applied yield load solutions for both, maximum crack size at boost loading and final failure are probably conservative. However, even moderate changes in the yield load could be of tremendous effect on the final result.**
- **Therefore, a throughout validation of the recently proposed yield load solutions and, perhaps, the development of further improved solutions is a key issue for reducing the inherent conservatism of fracture mechanics analyses in the context of an overload philosophy.**

Further improvement possible by applying a „reference load“ instead of a limit load F_Y

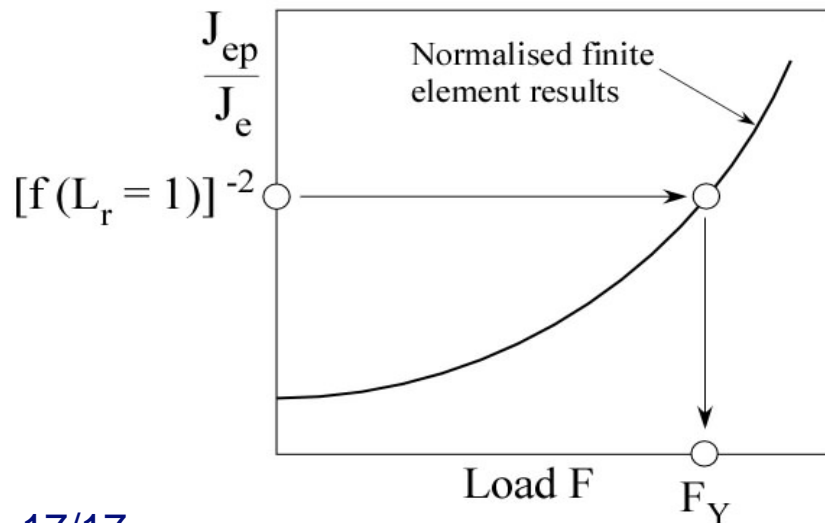


Overview presentation of Zerbst & Schwalbe (this workshop)

For $L_r = 1 \implies \sigma_{ref} = \sigma_Y \implies$ specific value of $f(L_r)$ dependent on the material

Proposal

$$f(L_r) = \begin{cases} \left[\frac{E \cdot \epsilon_{ref}}{\sigma_{ref}} + \frac{1}{2} \cdot \frac{L_r^2}{(E \cdot \epsilon_{ref} / \sigma_{ref})} \right] & \text{for } 0 \leq L_r \leq L_r^{max} \\ 0 & \text{for } L_r^{max} = \frac{\sigma_Y + R_m}{2\sigma_Y} \end{cases}$$



Defines a specific ratio of J/J_e corresponding to F_Y

$$\frac{J}{J_e} = [f(L_r)]^{-2}$$

$F_Y =$ „Reference load“