

Fracture control on launchers: Industrial Implementation

ESA Workshop on Fracture Control of Spacecraft, Launchers and
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Astrium Space Transportation

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All the space you need

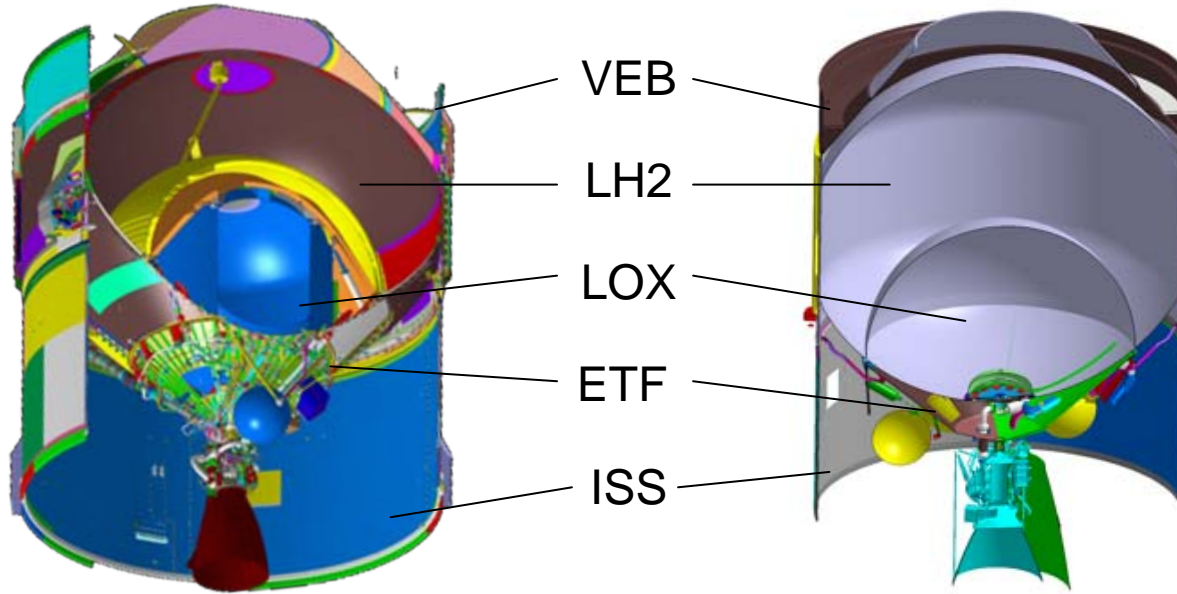


Outline

- Current and upcoming developments
- Fracture Control requirements and approaches
- Comparison of Fracture Control approaches
- Standards SG-1-X-10 and ECSS-ST-32-01
- Organization of Fracture Control activities
- Industrial experience
 - Critical design issues
 - Critical production issues and experience
- Conclusions and Outlook

Current and upcoming developments

ESC-A



Ariane 5 ME Upper Stage „Crius“

- Points critical for fracture control:
 - Generally: geometry, load, material
 - Specifically: location, wall thickness, load type, material, SCC sensitivity
- pressurized welds, rings, lines, covers, (some) brackets

Fracture control requirements (ESC-A)

■ Applicable documents:

- A5-SG-1-X-10-ASAI: Structure design, dimensioning and test specifications (proprietary)
- A5-NT-13-X-0008-DASA: Design criteria for ESC-A (proprietary)
 - Contains dedicated requirements for dimensioning and DTA, on top of SG-1-X-10
- A5-NT-1e6-H-539-CNES: DTA rules for A4 re-flown H/W (propr.)
 - Contains damage tolerance requirements for re-used Ariane 4 components
- ESA-PSS-01-401: ESA Fracture Control Requirements

■ Reference documents:

- ECSS-E-30-01A: Fracture Control
- ECSS-Q-70-36A: Material selection for controlling SCC
- MIL-STD-1522A: Pressurized missile and space systems
- ANSI/AIAA S-080 + 081: Pressure vessels metallic + COPVs

Fracture control approaches

- ESC-A stage was developed to SG-1-X-10
 - For design issues not contained there ECSS-E-30-01A has been considered
- Crius stage will be developed to the current SG-1-X-10 (same as for ESC-A)
 - no significant differences in the basic approach
 - For design issues not contained there ECSS-ST-32-01A will be considered

Main issues:

- No PFCIs, FCIs but DJ (~FC Summary), NTs → FC traceability can be optimised
- Safe-life in linear-elastic analysis always required to SG
- Fail-safe analysis performed to ECSS
- Elastic-plastic analysis: not regulated
- SCC requirements (updated data with ECSS, in-house results)
- Only few minor or negligible hazard items, only minimum low risk items

Standard SG-1-X-10

- **Current SG-1-X-10 partly obsolete**
 - No rules related to structural screening (managed only via RAMS)
 - Only linear-elastic approach, no fail-safe or low risk, SCC very general reqs.
 - No position on material constitutional laws (→ damage mechanics!)
 - Few documentation requirements (essentially NT,DJ)
- **New SG-1-X-10 (in draft)**
 - Comprises all aspects of structural dimensioning, incl. damage tolerance
 - Defines requirements for both structures and propulsion
 - Most points related to damage tolerance improved (fail-safe / low-risk classifications, points on EPFM, documentation)
 - Classification comparable to ECSS, but less sophisticated logic
 - Less detailed requirements on justification performance and NDI than in ECSS
 - Will not be applicable for Crius, but some features from it will be implemented via dedicated additional requirements

Standard ECSS-ST-32-01

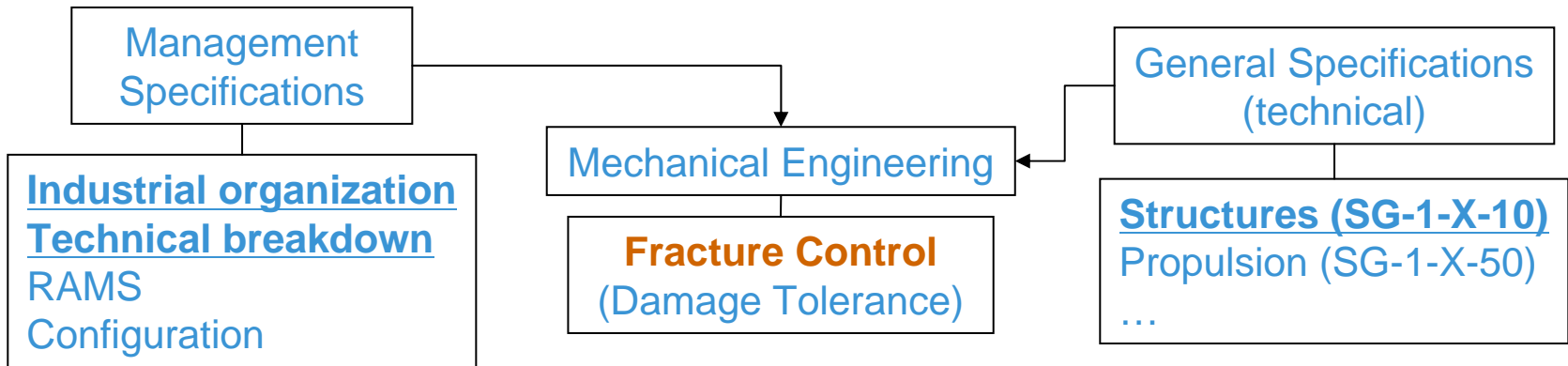
■ ECSS-E-ST-32-01

- Clean instructions on screening and classification
- Detailed requirements on justification methods (though little on EPFM)
- Detailed requirements on NDI, clearly linked to the sensitivity limits
- Dedicated requirements on particular parts / technologies (welds, castings, fasteners, composites)
- Special attention paid to quality assurance / traceability
- Low risk approach: sensitive issue in view of production stability

■ ECSS and ESC-A / Crius

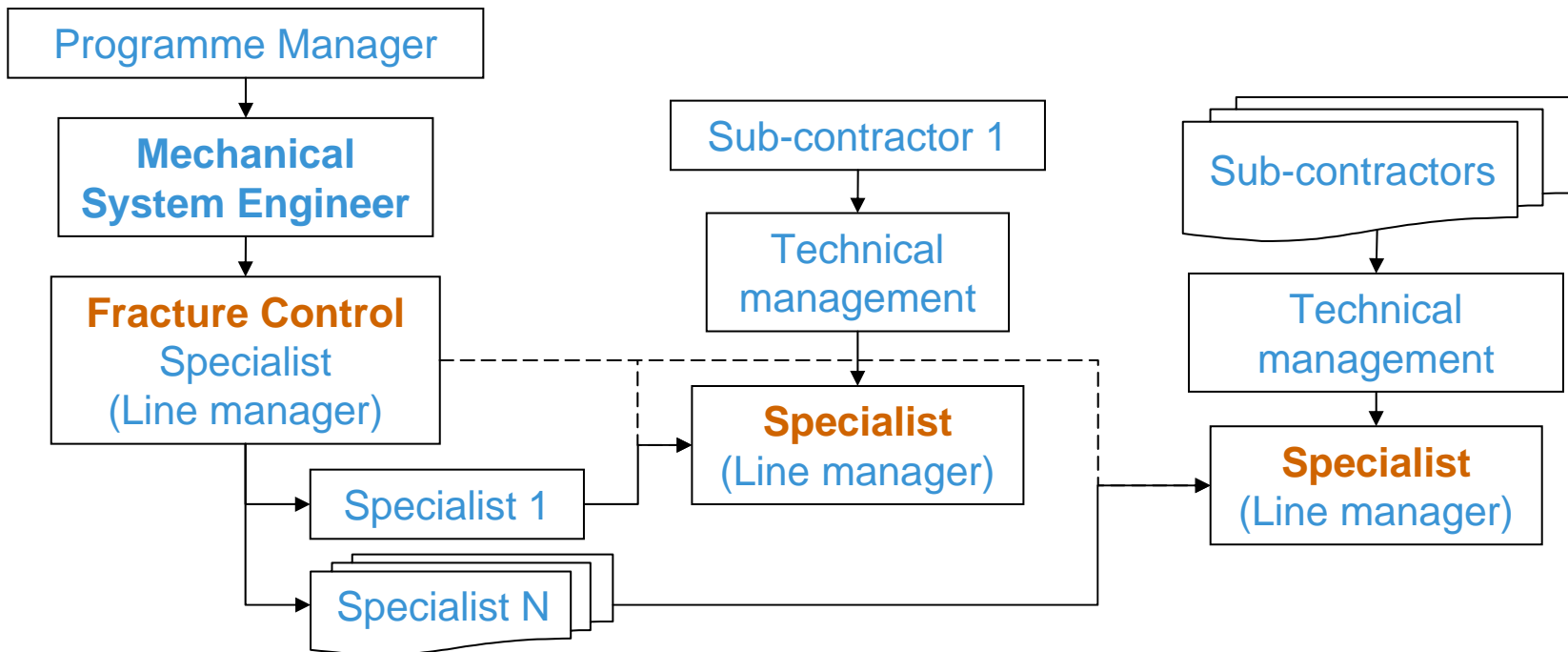
- Direct application difficult: ECSS is too inconsistent with A5 industrial rules, (arrangement of FC activities)
- Technically, most of the contents are directly transferable
- ECSS is related strictly to Fracture Control – SG-1-X-10 comprises the complete dimensioning
- ECSS is related primarily to fracture control of structures – new SG-1-X-10 includes also propulsion

General environment (development phase) Organization of Fracture Control activities (1)



- Mechanical Engineering activities are regulated via several management specification (SMs)
 - Industrial organization → requirements on technical consistency
 - Technical breakdown → arrangement of engineering
- Mechanical Engineering activities are bound to a certain item and part of the so called constitution list
- Contrary to the NSTS rules: fracture control is not so strongly related to Safety (like e.g. NASA-5019 to NSTS 1700.7), but is viewed as a part of overall structural justification

Practical management (development phase) Organization of Fracture Control activities (2)



■ Fundamental organizational tasks for Fracture Control:

- Definition of inputs: geometry, loads, preliminary NDI, preliminary material properties
- Review of outputs: results of DTA, material qualification, NDI qualification

Critical design issues (relevant for development phase)

- Material selection (fracture and crack growth properties, technological properties, SCC)
 - Sometimes obsolete materials (7020,7010)
 - Precise knowledge of the component's envelope life crucial
- Definition of wall thicknesses and local geometry (stress gradient, LBB)
- Reduction of the number of safe-life interfaces (too many safe-life bolts)
- Selection of NDI
 - NDI relevant for the potential defects, realistic assumptions on detection capability
- Feasibility of justification by test
 - When analysis not meaningful, e.g. due to highly complex geometry and/or excessive loading/plastification
- Proper modelling
 - Implementation of new Esacrack/Nasgro versions vital, but sensitive
- Applied elastic-plastic methodology
 - Current SG-1-X-10 does not regulate this issue; ECSS does not give any details on EPFM
 - Further industrialization of damage mechanics models (Rice&Tracey, Gurson, ...)
- Robustness of justifications
 - Early and proper identification of relevant processes

Critical production issues (relevant for development and production phases)

- **Material production**
(casting, working/rolling)
 - Microstructure similarity, texture
 - Chemical purity and homogeneity (and its reproducibility)
 - Mechanical properties isotropy (homogeneity in always one direction)

- **Material processing**
(forming, machining)
 - Proper design of the process (relevance of tolerance fields versus over/underthicknesses)
 - Complex fragmented areas → “fail-safe” milling machines desirable

- **Welding and NDI**
 - Proper design of the process (reasonable elimination of human factor)
 - Maximum vigilance: with fusion welding human factor especially important, including NDI
 - Weld repairs are always critical: stability of the nominal welding is crucial

- **General quality**
 - Traceability (running protocols → family building)

- **Objectives**
 - **To keep the launcher in the authorized (and well-proven) qualification range**
 - To avoid difficult damage tolerance justifications above the defect catalogue (→ robustness)

Fracture Control in future developments

- Further development of the technical basis
 - Consolidation of load spectra (clear I/F to subcos)
 - Early analysis of potential defects (esp. type, origin, size)
 - Unified NDI procedure for evaluation of sensitivity limits
 - Unified standard for SCC testing under propellants
 - Thorough understanding of materials and process characterization
- Particular points:
 - Implementation of Al-Li alloys
 - Implementation of new welding technologies, esp. FSW
 - Challenge in terms of reproducibility (both base material properties and processes, esp. welding and NDI)
- **Maintaining the validity of Fracture Control justifications may be harder than their initial achievement**

Conclusions and Outlook

- **By its nature Fracture Control takes special responsibility as it closes the loop between dimensioning and the real flight H/W**
 - Despite good design, proper dimensioning and conscientious quality unexpected problems may always occur in the production phase
 - Elimination of human factor has great quality importance both in Development (robustness of justifications) and Production processes (reproducibility)
 - **Where elimination of human factor is not reasonable detailed instructions are crucial**
- **In the first instance the importance of reasonably detailed and clean instructions applies for standards like ECSS-E-ST-32-01 or SG-1-X-10**

Most challenging for future developments

- Perform comprehensive analysis of all relevant production events and of the envelope life already during the development phase
 - Early and deep analysis of production processes
 - NDI relevance and sensitivity levels
 - Further implementation of elastic-plastic methodology
- further increase of robustness and flexibility of generic justifications (positive cost factor for the product life)